

Airworthiness Directive Schedule

Aeroplanes

Beechcraft D18S (Expeditor)

31 August 2000

- Notes**
1. This AD schedule is applicable to Beechcraft D18S aircraft, manufactured under Federal Aviation Administration (FAA) Type Certificate No. A-765.
 2. The date above indicates the amendment date of this schedule.
 3. New or amended ADs are shown with an asterisk *

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DCA/B18/1 Airworthiness Directive Compliance at Initial Airworthiness Certificate Issue**Applicability:** All Model D18S**Requirement:** Compliance with the following FAA Airworthiness Directives (as applicable) is required:-

- 47-33-05 - Horizontal stabilizer spar
- 47-33-06 - Stabilizer spar replacement
- 47-33-07 - Static selector valve
- 47-34-01 - Expansion type filler cap
- 47-34-02 - Tail cone drain holes
- 47-34-03 - Fuel cell seal
- 48-13-01 - Generator control box
- 48-34-01 - Stabilizer attachment fittings
- 49-29-02 - Rudder spring bushings
- 59-18-02 - Fuel line modification
- 65-03-01 - Gap seal strips
- 66-30-01 - Static pressure line
- 68-23-06 - Volpar tri-gear
- 72-16-02 R3 - Volpar nose gear fork
- 72-25-06 - Nose gear extension
- 75-27-09 R2 - Wing structure
- 77-19-07 - Control column assembly
- 78-10-01 - Lower rudder torque tube

Note: Each part of this AD (each individual FAA AD) shall be certified in the aircraft log book separately.

Compliance: Before issue of New Zealand Airworthiness Certificate. Repetitive inspections to be accomplished at intervals not exceeding the times specified in the FAA Airworthiness Directives**Effective Date:** 7 May 1999*** DCA/B18/2 Cancelled**