

Airworthiness Directive Schedule

Helicopters

Leonardo AW169

30 September 2021

- Notes:**
1. This AD schedule is applicable to AW169 helicopters manufactured by Leonardo S.p.A. (formerly Finmeccanica Helicopter Division, AgustaWestland) under European Union Aviation Safety Agency (EASA) Type Certificate No. R.509.
 2. The European Union Aviation Safety Agency (EASA) is the National Airworthiness Authority (NAA) responsible for the issue of State of Design Airworthiness Directives (ADs) for these helicopters. State of Design ADs can be obtained directly from the EASA website at <http://ad.easa.europa.eu/>
 3. The date above indicates the amendment date of this schedule.
 4. New or amended ADs are shown with an asterisk *

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2017-0137 Canceled – EASA AD 2018-0203 refers**Effective Date:** 27 September 2018**2017-0155 Emergency Exit Windows – Inspection****Applicability:** Leonardo AW169 helicopters, S/N 69007, 69009, 69011 through to 69019 inclusive, 69021 through to 69024 inclusive, 69027, 69032, 69033, 69041, 69045 and 69051.**Compliance:** Initial compliance required before the issue of a New Zealand Certificate of Airworthiness, or at the next Review of Airworthiness (RA), whichever is the sooner, unless previously accomplished. Repetitive inspections, if required, are to be accomplished at intervals not to exceed the times specified in the EASA AD.**Effective Date:** 22 February 2018**2017-0156 Automatic Flight Control System – Inspection****Applicability:** AW169 helicopters, all S/N except those helicopters embodied with Automatic Flight Control System (AFCS) software P/N 6F2210AS0103 or later.**Compliance:** Initial compliance required before the issue of a New Zealand Certificate of Airworthiness, or at the next Review of Airworthiness (RA), whichever is the sooner, unless previously accomplished. Repetitive inspections, if required, are to be accomplished at intervals not to exceed the times specified in the EASA AD.**Effective Date:** 22 February 2018**2017-0164 Landing Gear Retraction Actuators – Inspection****Applicability:** AW169 helicopters, S/N 69007, 69009, 69011, 69013, 69014, 69015, 69017, 69018, 69020, 69021, 69022, 69023, 69024, 69025, 69027, 69028, 69031, 69032, 69041, 69042, 69043, 69044, 69049 and 69051, and

AW169 helicopters, all S/N fitted with a retractable landing gear system P/N 6F3200F00311 or 6F3200F00411.

Compliance: Initial compliance required before the issue of a New Zealand Certificate of Airworthiness, or at the next Review of Airworthiness (RA), whichever is the sooner, unless previously accomplished. Repetitive inspections, if required, are to be accomplished at intervals not to exceed the times specified in the EASA AD.**Effective Date:** 22 February 2018**2017-0255 MLG Weight on Wheels Support Bracket – Inspection****Applicability:** AW169 helicopters, all S/N fitted with a LH MLG assembly P/N 6F3210V00132 or 6F3210V00133 and a S/N from MN01 through to MN84 inclusive, and/or a RH MLG assembly P/N 6F3210V00232 or 6F3210V00233 and a S/N from MN01 through to MN81 inclusive, except if the MLG has Magnaghi Aeronautica SB-07-2017-AW169 embodied.**Compliance:** Initial compliance required before the issue of a New Zealand Certificate of Airworthiness, or at the next Review of Airworthiness (RA), whichever is the sooner, unless previously accomplished. Repetitive inspections, if required, are to be accomplished at intervals not to exceed the times specified in the EASA AD.**Effective Date:** 22 February 2018

2018-0203 Pilot Collective Stick Assembly – Inspection

Applicability: AW169 helicopters, all S/N fitted with a Pilot Collective Stick Assembly P/N 6F6711A07832, or P/N 6F6711A07831.

Effective Date: 27 September 2018

2018-0197 Emergency Exit Windows – Inspection

Applicability: AW169 helicopters, all S/N and AW189 helicopters, all S/N, except those helicopters fitted with emergency exit windows with strap P/N A487A003A, or helicopters with bubble windows P/N 8G5620F00112.

Effective Date: 27 September 2018

2018-0208 Emergency Flotation System (EFS) – Inspection

Applicability: AW169 helicopters, all S/N fitted with an EFS float assembly, having any P/N.

Effective Date: 27 September 2018

2018-0252-E Tail Rotor Flight Control System – Inspection

Applicability: AW169 helicopters, all S/N.

Effective Date: 22 November 2018

2018-0261-E (correction)Cancelled – EASA AD 2019-0023 refers

Effective Date: 5 February 2019

2019-0019 Tail Rotor Actuator Input Lever – Inspection

Applicability: AW169 helicopters, all S/N.

Effective Date: 13 February 2019

2019-0022 (correction) Tail Rotor Duplex Bearing – Inspection

Applicability: AW169 helicopters, all S/N.

Effective Date: 2019-0022 - 5 February 2019
2019-0022 (correction) - 5 February 2019

2019-0023 Cancelled – EASA AD 2019-0121 refers

Effective Date: 3 June 2019

2019-0121R1 Cancelled – EASA AD 2019-0193 refers

Effective Date: 14 August 2019

2019-0193 Cancelled – EASA AD 2020-0048 refers

Effective Date: 20 March 2020

*** 2019-0252 Cancelled – EASA AD 2021-0199 refers**

Effective Date: 30 September 2021

2020-0013 Hand Held Fire Extinguishers – Inspection

Applicability: AW169 helicopters, all S/N.

Effective: 27 February 2020

2020-0048 Cancelled – EASA AD 2020-0197 refers

Effective: 24 September 2020

2020-0197 Tail Rotor Servo Actuator / Duplex Bearing – Inspection

Applicability: AW169 helicopters, all S/N.

Effective: 24 September 2020

*** 2021-0199 Yaw Pedal Assemblies – Modification**

Applicability: AW169 helicopters, all S/N.

Effective: 30 September 2021