# **Type Acceptance Report**

TAR 1/21B/1 HAI Y12 IV

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#### Introduction

This report details the basis on which Type Acceptance Certificate No.1/21B/1 was granted in the Standard Category in accordance with NZCAR Part 21 Subpart B.

Specifically the report aims to:

- (a) Record the airworthiness certification standard used for type acceptance of the applicable model in New Zealand;
- (b) Summarise any outstanding requirements that must be complied with for the issue of a NZ Airworthiness Certificate to any models covered by the Type Acceptance Certificate.

# Foreign Type Certificate Details

Type Certificate:	A00006WI	
Issued by:	Federal Aviation Administration	
Manufacturer:	Hafei Aviation Industry Company Limited	
Model:	Y12 IV	
Engines:	2x Pratt & Whitney Canada PT6A-27	
Propellers:	2x Hartzell HC-B3TN-3B/T10173(N)B-3	
MCTOW	12,500 lb (5670 kg)	
Noise Category:	FAR 36, including Amendments 36-1 through 36-20 [ <i>Takeoff noise level 82.9 db</i> ( <i>A</i> )]	

The certification basis of the Y12 IV is FAR Part 23 effective February 1, 1965, including Amendments 23-1 through 23-42. This is an acceptable certification basis in accordance with NZCAR Part 21B Para §21.41, as FAR 23 is the basic standard for Commuter Category Airplanes called up under Part 21 Appendix C. There are no non-compliances and no special conditions have been prescribed by the Director under §21.23.

# Type Acceptance Application

The application for New Zealand type acceptance was from the Australian agent for the Y12 aircraft, Aircraft Technical Marketing, dated July 2000.

Type Acceptance Certificate No.1/21B/1 was granted on 24 November 2000.

The Yunshuji-12 (transport aircraft 12) is an enlarged development of the Y-11, a 10-11 seat piston twin light transport powered by two AI-14 radial engines. The Y-12 prototype first flew in 1982, and there have been three series production versions:

- Y12 (I) Initial version with 500 shp PT6A-11 engines, 93" diameter propellers, leading edge slats, seating for 17 passengers and a MTOW of 5000 kg.
- Y12 (II) Later production version with higher-rated 600 shp PT6A-27 engines, plain leading edge, 98" diameter propellers and MTOW increased to 5300 kg. The UK CAA issued Type Certificate FA49 for this variant in 1990 and examples are in operation in Malaysia and Fiji.
- Y12 (IV) Improved model with sweptback wingtips, redesigned seating for 18-19 passengers and max. weight increased to 5700 kg. The Canadian Aerospace Corporation markets a version of the aircraft under the Twin Panda name as a DHC-6 replacement.

The Y12 was originally produced by Harbin Aircraft Manufacturing Corporation (HAMC), and under Chinese commercial arrangements is marketed by the China Aviation Technology Import and Export Company (CATIC). HAMC has subsequently been reorganised in 1999 as Hafei Aviation Industry Co. Ltd (HAI), responsible for the Y12 and other sub-contract work.

# Type Data

The type data requirements of NZCAR Part 21B Para §21.43 have been satisfied by supply of the following documents:

- (1) Type certificate: FAA TC No. A00006WI for Harbin Y12 IV issued March 25, 1995 FAA TCDS No. A00006WI at Revision 3 dated July 16, 1996
- (2) Airworthiness design requirements: Already held by the CAA
- (3) Certification compliance listing:

Type Qualification for Y12IV Aircraft – Conformity Check List – April 23, 1994 Flight Test Report No. Y12IV-FB-1051 – Takeoff Noise Certification Test HAMC Report Y11T-JS-413 – Fatigue Life Substantiation – Dec.2, 1988

(4) Flight manual: HAMC Y12 IV Pilot's Operating Handbook and CAAC Approved Aircraft Flight Manual – Y12 IV SJW1 – CAA Accepted as AIR

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- (5) Illustrated Parts Catalogue: HAMC Y12IV Illustrated Parts Catalog – Y12 IV SJW7 (Volumes 1-5)
- (6) Maintenance manual and service data for aircraft, engine and propeller: HAMC Y12IV Aircraft Maintenance Manual – Y12 IV SJW4 (Volumes 1-5)

Maintenance Data already held by the CAA for the engine and propeller

- (7) Agreement from manufacturer to supply updates of data in (4), (5) and (6):Letter from Hafei Aviation Industry Co. Chief Designer dated May 18, 2000.
- (8) Other information:

HAMC ZHZ-904 – Control Procedure for Corrective and Preventive Actions HAMC ZHZ-2002 – After-Sales Service Control Procedure for Delivered Aircraft HAMC ZHZ-2004 – Preparation and Accomplishment of Aircraft SB and SL HAMC ZHZ-2005 – The Disposal Procedure of Field-Feedback Quality Information

HAMC ZHZ-2106 - Report Management Procedure of Fault, Failure and Defect

#### **Additional New Zealand Certification requirements**

Compliance with the following additional NZ requirements has been reviewed and were found to be covered by either the original certification requirements or the basic build standard of the aircraft, except as noted:

#### **Civil Aviation Rules Part 26**

#### Subpart B - Additional Airworthiness Requirements

Appendix B - All Aircraft

PARA:	REQUIREMENT:	MEANS OF COMPLIANCE:
B.1	Marking of Doors and Emergency Exits	FAR §23.811(b) for Commuter Category
B.2	Crew Protection Requirements - CAM 8 Appdx. B # .35	Agricultural Aircraft – Not Applicable

Appendix C - Air Transport Aircraft - More than 9 Pax

PARA:	REQUIREMENT:	MEANS OF COMPLIANCE:
C.1	Doors and Exits	FAR §23.807(b) and §23.807(d)(2)
C.2.1	Additional Emergency Exits - per FAR 23.807(b) @ 10.5.93	FAR §23.807(d)(1)(ii) – Main passenger/cargo door on the
	For 16-23 passengers must have one exit on the same side as	port side at Fr.16-20, Emergency exits on either side at Fr.9-
	the passenger door and two exits on the opposite side.	11 and one at Fr.16-18 on the starboard side.
C.2.2	Emergency Exit Evacuation Equipment – Descent means	FAR §23.807(d)(1) – Y12 exits less than 2m from the ground
C.2.3	Emergency Exit Interior Marking - Size/self-illuminating	FAR §23.811(b) for Commuter Category
C.3.1	Landing Gear Aural Warning - Automatic Flap Linking	Not Applicable – Fixed undercarriage

# **Civil Aviation Rules Part 91**

#### Subpart F - Instrument and Equipment Requirements

PARA:	REQUIREMENT:		MEANS OF	COMPLIANCE:
91.505	Shoulder Harness if Aerobatic; >10 pax; Flight Training		IPECO pilot seat includes sho	oulder belt – see Flight Manual §
			7.15	
91.507	Pax Information Signs - Smoking, safety belts fastened		Distributor advises these are f	fitted as standard
			To be determined on an indi	vidual aircraft basis
91.509	(1) ASI	FAR §23.1303(a) * #1	(8) Coolant Temp	N/A - Turbine engined
Min.	(2) Machmeter	N/A	(9) Oil Temperature	FAR §23.1305(c) * #19
VFR	(3) Altimeter	FAR §23.1303(b) * #3	(10) Manifold Pressure	N/A - Turbine engined
	(4) Magnetic Compass	FAR §23.1303(c)	(11) Cylinder Head Temp.	N/A - Turbine engined
	(5) Fuel Contents	FAR §23.1305(a) * #21	(12) Flap Position	FAR §23.699(a)(2) * #27
	(6) Engine RPM	FAR §23.1305(d)(e) * #18	(13) U/c Position	N/A – Fixed undercarriage
	(7) Oil Pressure	FAR §23.1305(b) * #19	(14) Ammeter/Voltmeter	FAR §23.1351(d) – Both fitted as
	* Fitted as standard – see	Flight Manual Fig. 7-9a		standard – see FM Fig.7-9a #47
91.511	(1)Turn and Slip	Fitted as std –FM Fig. 7-9a #6	(3) Anti-collision Lights	FAR §23.1401 – See FM §7.27
Night	(2) Position Lights	FAR §23.1385 – see FM §7.27	(4) Instrument Lighting	FAR §23.1381 – See FM §7.27
91.517	(1) Gyroscopic AH	Fitted as std – FM Fig. 7-9a #2	(5) OAT	Fitted as std – FM Fig.7-9a #10
IFR	(2) Gyroscopic DI	Fitted as std – FM Fig. 7-9a #7	(6) Time in hr/min/sec	Fitted as std – FM Fig.7-9a #31
	(3) Gyro Power Supply	FAR §23.1331(a)(3)	(7) ASI/Heated Pitot	Fitted as std – see FM §7.29
	(4) Sensitive Altimeter	Encoding altimeter 101420-	(8) Rate of Climb/Descent	VSI fitted as standard – see
		11934 fitted as standard		Flight Manual Fig.7-9a #8
91.519	IFR Communication and Navigation Equipment		Standard IFR installation incl	udes:
			Collins VHF-251 and HF-230	) radios;
			Collins ADF-650A ADF and	DME-451 DME;
			Collins VIR-351 VOR and M	KR-350 Marker Beacon;
91.523	(a) More Than 10 pax - First Aid Kits per Table 7		To be determined on an indi	vidual aircraft basis
Emergcy	- Fire Extinguishers per Table 8		2 hand-held fire extinguishers	s fitted as std – see FM

Eqpmt.	(b) More than 20 pax - Axe readily acceptable to crew	N/A – maximum 19 passengers
	(c) More than 61 pax - Portable Megaphones per Table 9	N/A – Less than 61 passengers
91.529	ELT - TSO C91a after 1/4/97 (or replacement)	To be determined on an individual aircraft basis
91.531	Oxygen Indicators - Volume/Pressure/Delivery	Operational requirement - To be determined as required
91.533	>30 min above FL100 - Supplemental for crew, 10% Pax	Not fitted as standard (Maximum operating altitude specified
Unpress.	- Therapeutic for 3% of Pax	in Flight Manual Limitations Section is 23,000 ft.)
A/c	Above FL100 - Supplemental for all Crew, Pax	Portable Oxygen System available as an option – see Flight
	- Therapeutic for 1% of Pax	Manual Supplement No.6
	- 1201 PBE for each crew member	
91.541	SSR Transponder and Altitude Reporting Equipment	Collins TDR-950 fitted as standard equipment [KT76A
		Optional]
91.543	Altitude Alerting Device - Turbojet or Turbofan	N/A – Turboprop powered
91.545	Assigned Altitude Indicator	Operational requirement - To be determined as required
A.15	ELT Installation Requirements	To be determined on an individual aircraft basis

#### **Civil Aviation Rules Part 125** Subpart F - Instrument and Equipment Requirements

PARA:	REQUIREMENT:		MEANS OF COMPLIANCE:
125.355	5 Seating and Restraints		FAR §23.785
125.357	Additional Instrum	ents (Powerplant and Propeller)	FAR §23.1305 at Amendment 42 effective February 4, 1991
125.359	Night Flight	Landing light, Pax compartment	Operational requirement - To be determined as required
125.361	IFR Operations	Speed, Alt, spare bulbs/fuses	<b>Operational requirement - To be determined as required</b>
125.361	SE IFR Requirements – If Applicable		N/A – Only applicable to single-engined aircraft
125.363	Emergency Equipment (Part 91.523 (a) and (b))		To be determined on an individual aircraft basis
125.365	Public Address and Crew Member Intercom System		AUD-251A intercom fitted as standard – Passenger Address
			currently standard equipment on the Y12 II Model
			To be determined on an individual aircraft basis
125.367	Cockpit Voice Recorder		Fairchild A100A CVR available as optional fit – See FMS No.17
125.369	Flight Data Recorder		Not fitted as standard equipment or presently available as an option
			To be determined on an individual aircraft basis
125.371	Additional Attitude Indicator		N/A – Turboprop powered
125.373	Weather Radar		N/A – MCTOW less than 5700 kg. [RDR-1400C optional fit]
125.375	Ground Proximity Warning System		N/A – MCTOW less than 5700 kg.
125.377	HUMS		N/A – Only applicable to single-engined aircraft

# Summary

Type Acceptance Certificate No. 1/21B/1 has been granted to the Harbin Model Y12 IV and serial numbers 008 and up, per the FAA type certificate, are now eligible for the issue of a New Zealand Airworthiness Certificate in the Standard Category in accordance with CAR §21.177, subject to any outstanding operational requirements noted above being met.

# Attachments

The following documents form attachments to this report:

Three-view drawing Harbin Model Y12 IV Copy of FAA Type Certificate/Type Certificate Data Sheet A00006WI Sign off

David Gill Team Leader Airworthiness

Date: 24 November 2000